Fig.

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20 Arn 1958

FOURTH ENDORSEMENT on NAAS Kingsville AAR ser 13-57 concerning TV-2 BuNo 138043 accident occurring 17 Jul 57

From: Chief, Bureau of Aeronautics

To: Chief of Naval Operations (OP-57)

Via: Commander, U.S. Naval Aviation Safety Center

Subj: Aircraft Accident Report; forwarding of

Ref: (a) Telcon between CNO (OP-57) and Chief, Bureau of Aeronautics (MA-61) of 14 April 1958

- 1. By reference (a) an agreement was reached for the forwarding of certain Aircraft Accident Reports without engineering comments.
- 2. The basic report with subsequent endorsements meets the criteria of reference (a) and is forwarded less comment.





By direction .

SPECIAL HANDLIFF HERRINGO THE ACCUMENTED WITH PART VIL CHMATTEST 1980.68

THIRD ENDORSESSET on Assortingeville Ask ser 13-57, concerning TV-2, BURG 138043, opens and W Joly 1962, Plant BASKINS

From Chief of Neval/Air Training

To: Chief of Mary Coperations (Op-57)

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(2) Trector, U. S. Maval Aviation Safety Canter

Subja Aircraft Accident Paport; farwarding of

1. Forwards a concurring in the conclusion and recommendations of the Board as willied by the second endorsement. Concur in the setting and comments whiseled in the first endorsement.

(b) (6

by direction

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CO NAAS KINGSVILLE (4)

"Special Handling Required in Accordance with Part VII OPNAVINST 3750.6B" 125/AAR 5 SEP 1957 SECOND ENDORSEMENT on NAAS Kingsville AAR Ser 13-57, concerning TV-2, BuNo 138043, accident occurring 17 July 1957, pilot HASKINS Chief of Naval Air Advanced Training Chief of Naval Operations (OP-57) To: (1) Chief of Naval Air Training Via: (2) Chief, Bureau of Aeronautics (MA-61) (3) Director, U S. Naval Aviation Safety Center Aircraft Accident Report; forwarding of Subj: (a) MAS CorpC O&R LES No. 51-53 Addendum No. 29 of 29 July 1957 Ref: 1. Forwarded, concurring with the conclusions and with all but the first recomme dation of the Ajroraft Accident Board. The comments of the first endorsement are noted and concurred in with the exception of paragraph 2. which has reference to the Board's first recommendation, 2. Reference (a) has been promulgated to all TV-2 units of this Command and provides for the replacement of the existing transmitter float guard in the earlier series TV-2 airplanes and the addition of this improved transmitter float guard to the later series TV-2 aircraft. In addition, reference (a) required the replacement of the cylindrical type float of tank unit EA565W-553 to the ball type float of tank unit EA565W-1339. It is strongly believed that these improvement will eliminate erroneous indications of fuel quantity in the main fuselage tank, as has occurred too often in the past. 3. The Board's first recommendation has merit but is not concurred with because the changes outlined in reference (a), when incorporated, should provide reliable fuel quartity indications of the fuselage tanks. In addition, the changing to a capacitance type system, taking time to process an aircraft service change, available funds and age of the majority of the TV-2 airplanes into consideration, is not considered realistic. 4. A system of fuel management in which all fuel tank switches are gangbarred immediately after take-off vice manually controlling has been recommended to all TV-2 units of this Command, but as brought out by the Board's second recommendation and paragraph 2. of the first endorsement, this procedure has certain objections unless a fuel flow detector is provided to alert the pilot he is losing fuel through the overboard drain. 5. The comments of paragraph 4. of the first endorsement are highly concurred in. The latest information to this Headquarters is that the delivery of the

"Special Handling Required in Accordance with Part VII OPNAVINST 3750.6B"

NEL/05 | 5 SEP 1957

remaining quantities of 650 automatic parachute devices, allocated to CNAVANTRA by ASO Speedletter of 19 February 1957, are scheduled to commence during the second week of September 1957.

2 R Seale

F. R. DRAKE Chief of Staff

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5. This accident will be given wide dissemination.

Copy to:

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2CC USNAVAVSAFECEN

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V. THE ACCIDENT.

At 1937 on 17 July 1957, LCDR H. L. HASKINS, instructor pilot, and ENS A. P. BURROWS, student pilot, departed NAS Forest Sherman, Pensacola, Florida, in TV-2, BUNO 138043, under instrument flight rules to NAAS Kingsville, Texas. The flight was of a training nature, designated as a 3D22 (C-7) cross-country.

D. ARTC cleared the flight via direct to Saufley, Red 96 to New Orleans, Green 6 to Corpus Christi, Red 30 to Bishop Intersection, direct to Kingsville, and also cleared them to maintain "1,000 feet on top". Actual altitudes enroute are unknown, but at Palacios, San Antonio ARTC cleared the flight to descend to 20,000 feet from 35,000 feet. Initial contact with Kingsville Approach Control was made over Corpus Christi at a reported altitude of 20,000 feet. Arrival at the Riviera Radio Beacon was six minutes later, at 2123. Approach Control had cleared the flight to commence a Kingsville jet penetration from 20,000 feet, with no delay and the penetration was commenced on arrival, at 2123. At 21262, the student reported, "Inbound out of penetration turn," He reported Riviera inbound at 2130. Approach Control then gave instructions for the flight to: "Report Low-Frequency Range Station on final." The receipt of this transmission was acknowledged by the student. At 2132 the following transmission was received by Approach Control, "MAYDAY, MAYDAY, MAYDAY, Fuel Pressure!" and was identified from the tape recording as LCDR HASKINS' voice. The "MAY-DAY" was acknowledged by Approach Control who immediately asked if 8043 would work the tower on the present frequency. Either the tower or Approach Control then asked if he had passed low station. The only answer was "EIGHT ZERO", (determined from tape recording). The next transmission from 8043 was, "800 feet, no choice but a night flame-out landing." The ensuing explosion on impact was observed at approximately 2134.

VI. DAMAGE TO THE AIRCRAFT.

A. Strike damage was incurred by TV-2, BUNO 138043 as a result of impact with the ground, collision with numerous trees and fire. Initial contact was made by an undetermined part of the aircraft with two (2) wooden fence posts about four feet high and spaced approximately twelve feet apart. One post was broken off about two feet above the ground, the other lost about two inches from the top. This is the point of initial contact and all measurements will be taken from this point (Enclosures (8) and (9)). Ninety-six (96) feet further along a general flight path of 330° magnetic, a nine-inch diameter tree was struck by the port wing three feet above the ground and three feet inboard of the tip tank (Enclosure (10)). This section and the port tip tank parted from the aircraft. The tip tank was found 316 feet along, and eight feet to the right of the flight path. The severed section of the wing was recovered at 341 feet along and 25 feet to the right of the path of the aircraft. After striking this tree, the aircraft continued to disintegrate due to what was deduced to be a cartwheel or similar uncontrolled maneuver. Portions of

the port wing fuel cell and the port tip tank pressure lines were scattered along the flight path before the aircraft struck a second tree, 240 feet from initial contact and two feet above the ground, leaving a part of the port wing panel embedded and the outer half of the port aileron lying three feet to the left (Enclosure (11)). At 255 feet the inboard half of the port aileron was found. At 306 feet the aft section of the fuselage and the starboard horizontal stabilizer struck several limbs of another tree 12 feet above the terrain (Enclosures (12) and (13)). The aircraft was apparently in a right wing-high, tail-high attitude. A fourth tree, two feet in diameter, was contacted at a distance of 431 feet from initial contact with various components being scattered along the flight path over another 108 feet. Included in these components, and the order in which they were found, were the empennage (Enclosure (14)), port gunbay door.

- B. In this general area, also, and to the left of the flight path, a part of the nose section, the starboard tip tank (in two (2) pieces), and a very small portion of the port alleron with electrical trim tab attached, were found. The first signs of fire were indicated in this area by small burned patches. At a point 674 feet from initial contact a fifth tree was struck and a part of the starboard wing and flap were found lodged in the base of the tree. A considerable area around this tree was badly burned. Perpendicular and 25 feet left of this point, the canopy was found. Thirty-five feet further (709 feet from initial contact) the instructor pilot's helmet was recovered.
- C. It is believed that the engine and fuselage separated very near this point. An eleven foot section of the starboard wing was located at 769 feet and the engine came to rest at 829 feet from the point of initial contact. After the engine and the remainder of the fuselage separated, (Enclosure (15)), the fuselage veered in a curved path to the right. It came to rest against the base of a large tree on an approximate heading of 080°, due east of the engine. The windscreen assembly was recovered 25 feet beyond the tree (Enclosure (16)). The cockpits and forward portion of the fuselage were almost completely consumed by fire which made it impossible to ascertain final impact damage (Enclosure (17)).

VII. THE INVESTIGATION.

- A. The crash occured at 2134 as the explosion was observed by tower personnel.
 - B. Crash followed flame-out of undetermined cause.
- C. Landing was made downwind with landing gear and speedbrakes up, landing flaps down.
- D. Contact airspeed was estimated to be 130 knots, 20 knots, in a relatively flat attitude.

- E. The engine was not running on impact and suffered very little damage. It was examined by O&R and judged to be capable of proper operation. F. One inverter was examined by O&R and all indications showed it was operating on contact. G. The fuel transfer pumps were examined by O&R and this examination revealed only that they were capable of proper operation. H. The front and rear cockpit totalizers were also examined by O&R. A reading of 96 gallons remaining in the fuel tanks could be positively determined from the last, two (2) counter wheels, but it was impossible to extract a reading from the first wheel, I. The aircraft was fueled with 725 gallons of JP-4 at NAS Forest Sherman, With a capacity of 813 gallons, the possibility of not having a full load prior to the starting of the engine is very remote. J. Weather was not a contributing factor to this accident. However, it was a very dark night. K. Both pilots were physically qualified to participate on this flight. L. The canopy was jettisoned at a point 709 feet from initial contact. M. LCDR HASKINS was in the front cockpit and ENS BURROWS was in the rear cockpit. N. ENS BURROWS' ejection seat had been fired and was found in front of the port wing where the fuselage remains came to rest. O. The body of ENS BURROWS was found approximately eight (8) feet outboard of the fuselage and slightly aft of the remaining port wing. P. The review of the pilots discrepancy sheets indicate the following pertinent discrepancies and maintenance action taken:
 - 1. On 13 March 1957 the pilot discrepancy form reported: Fuselage tank still reads full, (upon interrogation, the pilot stated that the fuselage tank showed "FULL" after taxiing out and returning to the line). The following maintenance action was taken: "Changed float valve."
 - 2. On 25 May 1957 the pilot discrepancy form reported: Liquidometer stuck at full position, Maintenance action: "Unstuck and bent
 arm to prevent future sticking."

3. On 2 June 1957 the pilot discrepancy form reported: Fuse-lage tank stays full with the other tanks off. Maintenance action: "Unstuck liquidometer float. Turned up and checked out 0.K."

4. On 8 June 1957 the pilot discrepancy form reported: Stuck fuselage float. Maintenance action: "Unstuck fuselage float."

VIII. THE ANALYSIS.

A. Due to the fact that this accident occured at night in a rather isolated area, there were no witnesses who saw the aircraft contact the ground. It was determined from the tower operators observation and from the tower tape recording that the crash did occur at 2134. It was also determined from the tape recording that the pilot of TV-2, BUNO 138043 experienced fuel pressure difficulties and subsequent flame-out at or above 800 feet prior to the aforementioned crash.

B. The cause of the flame-out cannot be determined but could have resulted from any of the following possibilities:

- 1. Fuel control malfunction.
- 2. Fuel pump failure.
- 3. Inadvertent "stop-cocking".
- 4. The transferring of fuel manually and having a fuselage transmitter float stuck in the "UP" position.
- C. Considered in the order as presented above, the following opinions are derived:
- 1. The fuel control was recovered and examined by Corpus Christi
 O&R and found to be capable of proper operation.
- 2. The fuel pumps were also examined by O&R and it was revealed that they were capable of proper operation prior to the crash. Even if the primary element of the fuel pump had failed, the fuel flow would have been maintained by the secondary element, which would have eliminated a flame-out.
- 3. Inadvertent "stop-cocking" cannot be completely ignored, but the experience of both pilots in TV type aircraft make this possibility remote.
- 4. The transferring of fuel manually plus having a fuselage transmitter float stuck in the "UP" position is considered another possible cause of the flame-out. Most pilots prefer to transfer manually on a cross-country flight to avoid a possible loss of fuel due to a malfunction of the fuel float system.

D. If LCDR HASKINS had been transferring manually and had not gangbarred the fuel switches, a flame-out would have occurred at a 258-gallon reading on the totalizer. However, the readings on both fuel totalizers which were sent to Corpus Christi O&R for examination revealed. a reading of 496 gallons.

- E. Upon examination of LCDR HAKINS' flight log (enclosure (3)) which was submitted to the clearing officer at NAS Forest Sherman, it was evident that he intended to fly at 35,000 feet. At Palacios, LCDR HASKINS requested and received a clearance to descend from 35,000 feet to 20,000 feet. Therefore, it is the Board's assumption that he did fly at this altitude even though he asked for and received a "1,000 feet on top" clearance. Fuel data on the flight log indicated that he would arrive at the Riviera Radio Beacon with 320 gallons of fuel remaining out of an estimated 780 gallons at take-off. From the Riviera Radio Beacon "high station" to the time of flame-out, enough fuel would have been consumed to give a reading of 296 gallons on the totalizers, however, a 296 gallon reading would preclude the possibility of airframe fuel system failure or mismanagement.
- F. The possibility also exists that the totalizer readings could have been 196 gallons. In order to have flamed-out at this reading, the aircraft would have had to consume 100 additional gallons of fuel over preflight computations, at least 62 of which had to be taken from the leading edge and/or wing group. Due to LCDR HASKINS' limited cross-country instructional experience, taxi distance to take-off end of runway, delay for climb-out clearance, and type of climb-out clearance received, it is reasonable to assume that this additional fuel could have been consumed.
- G. From 13 March 1957 to 8 June 1957, this aircraft had a discrepancy history of the fuselage transmitter float sticking in the "UP" position. This would indicate a full fuselage tank at all times fuel or no fuel. If this were the case, the aircraft could have flamed-out without any indication on the fuel low level warning light, which should have gone on when 87 gallons were remaining. Since this light is located just above the airspeed indicator, and is the type which cannot be dimmed, it is reasonably safe to assume that had it gone on, at least one of the pilots would have observed it.
- H. Since fire destroyed most of the aircraft, and left but little tangible evidence, the Board could come to no conclusion as to the probable cause of the flame-out.
- I. Both pilots utilized lap belts, shoulder harness, and protective helmets, but impact severity was so great that this equipment was useless.

IX. CONCLUSIONS AND RECOMMENDATIONS. A. Conclusion: 1. The primary cause of the accident is undetermined. B. Recommendations: 1. Replace the present fuselage fuel quantity indicator with a capacitance type. It is believed this would be more reliable, and would also conform with other jet aircraft fuel calibration, i.e. pounds rather than gallons. 2. The installation of a fuel flow detector in the overboard drain. This would give pilots more confidence in the present TV fuel system, and promote the habit of "Gangbarring" the fuel switches. The Lockheed representative stated that this installation would require only a few man hours and be relatively inexpensive. 3. It should again be brought to the attention of all pilots that . the yellow sheets should be thoroughly analyzed prior to each flight, and that the aircraft should not be accepted for flight if maintenance corrective action seems to be inadequate for any listed discrepancy. 4. Whether or not a wheels-down landing would have lessened the damage to this aircraft could not be determined. However, it is still recommended to land wheels-down on any emergency landing.

d. This report shall be filed in the event of an aircraft accident/ incident which involves one or more of the following:

Death Ditching Injury, Bail-out or Ejection (attempted or successful) Therever physiological or psychological factors are involved Aircraft Ground Accidents resulting in mericus injury 2. Completion of the form shall be the responsibility of the flight

surgeon 3. For type socident and damage code refer to OFNAY INSTRUCTION 3750.64.

· This form shall be prepared in quadruplicate. One co

turned over to the Aircraft Ac	cident Board (or the Survival	and other	nterested individuals)	squouren jirght	The second
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		13.	ACCIDENT DESCRIPTION	MICHAEL REPORT OF THE PARTY OF	THE PERSON NAMED	Water Street,

INCLUDE HERE A PARAGRAPH GIVING A BRIEF BUT FACTUAL ACCOUNT DESCRIBING IMPACT, SPECD ON IMPACT, ATTITUDE ON IMPACT, ETC. ATTACH PROTOGRAPHS I

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or insunits (Dire opinion no to cause of such injury indicated above, "Undetermined" with an applification will not be accepted. See additionable to accepted. See additionable to accepted.

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1: Use separate form for each person. 2. Under layory Class, was following pay:

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lass than critical but definituly respiring five cropped applicalization
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listed in this spingery which result in death within 30 days shall be reported by letter to the original addresses.

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Standard Form 500 Promulgated August 1948 By Buheau of the Wedget Chroulet A-52

(COPT)

AUTOPSY PROTOCO

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(b) (6)	ASSISTANT	I					

CLINICAL DEAGNOSIS (Including operations)

CLINECAL RECORD

DOA.

Aircraft accident five miles S. S. of MAAS, Kingsville, Tume, in a TVD jet plane.

Fromen sections and body fluid cent to Aviation School of Medicine, Randelph Field, Texas, and to AAMIL, BANG, Ft. Sam Houston, Texas, for toxicological studies for general unknown. Was sent by plane 7-19-57.

PATHOLOGICAL DIAGNOSES

(b) (6)

20

APPROVED SIGNATION (D)

(b) (6)

LT NO WAR

457-31

BURRUIG, Alfred Porrow, El

(b) (6)

DISTRIBUTE

REGISTER NO.

Horgus

(NAME OF HOSPITAL OR OTHER MEDICAL FACILITY)

10-65216-L D. S. MANAGEMENT PROPERTY DATE:

AUTOPSY PROTOCOL

U. S. MAYAL HOSPITAL COMPUS GRRISTI, THEAS (COPT) MERCES, Alfred Porrow, 5 EXTERNAL EXAMINATION this was the body of a well developed, well nourished, white male, about 25 years of age, (b) (6)
hair was close-cropped and reg-brown. (b) (6)
(b) (6) INTERNAL EXAMINATION (b) (6)

(b) (6)

MURROUS, Alfred Perrow, 285, (b) (6) Sungar (b) (6) 3pleans (b) (6) (b) (6) Abdominal lymph modes: (b) (6) Livers (b) (6) Gallbladders (b) (6) Extrahepatic biliary passages: (b) (6) cooreas: (b) (6) Adrenal glander (b) (6) Gastro-intestinal tracts (b) (6) Kidneys: (b) (6) bladder (b) (6) Skeletons (b) (6) Mana Leas (b) (6)

AUTOPSY PROTOCOL

DATE AND HOUR DIED 7-27-57

2135

7-18-57 1115 ASSISTANT

CHECK ONE HEAD ONLY YRUNK DNLY FULL AUTOPSY

LY ME BONR

CLINICAL DIAGNOSIS (Including specialing)

CLINICAL RECORD

DOA.

Aircraft conident, TV 2 jot plane, five miles S. E. of MAAS, Kingoville, Terms.

Specimens: Fromen tissue, bodyfluid and muscle tissue forwarded by a Seetheraft 920 on 7-19-57, to Aviation Medicine, Randolph Field, Tomas, and to GAML, BANG, Ft. San Houston, Texas, for texicological studies for general unknown and lastic asid determinations.

PATHOLOGICAL DIAGNOSES

- 1) Bultiple injuries, extrems.

LE SC USIG

HANGERS, HAVEY LOPPY, LCD

HE OF HOSPITAL OR OTHER MEDICAL PACILITY

AUTOPSY PROTOCOL

0.00 CORPUS CHRISTI, THAS

NL/1 A57-30 18 July 1957

HASKIRS, Barry Leroy, LCRR, (b) (6)

· PITRUAL PLANTATION

U. S. NAVAL HOSPITAL COMPUS CHRISTI, TRAAS M19-25-anh M1/1 A57-00 18 July 1997

MASKINS, EARLY LATOY, LIER, (b) (6)

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INTERNAL BUANCHATION

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18 July 1957 abdeninal lymph nodeer (b) (6) (b) (6) Gallbladders' (b) (6) Cotton-Hepatic biliary passages: (b) (6) Pancreas: (b) (6) Adrenales Gestro-intestinal treat; (b) (6) Kidoswar ery bladders (b) (6) (b) (6) Prostates (b) (6) Nuscles: See external examination. (b) (6) Skeletont (b)(6)

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STATEMENTS OF TOWER OPERATORS CONCERNING ACCIDENT TV-2, BUNO 138043

Navy 8043 was completing a Kingsville standard penetration, had reported Riviera inbound at 2130, was advised to report low frequency range on final---his transmission was "Roger", and at this time and up until now everything was normal.

At 2132 the following transmission was made by 8043 on Channel 5, 252.4:

"MAYDAY, MAYDAY, Flame-out."

At this time Mr. TICER called the tower on intercon to advise, as there was jet traffic in the pattern.

I called 8043 and asked him to work the tower on this frequency--and asked him if he had passed the range station, so I could get a better fix on him. 8043 came back with "Flame-out. No choice but night flame-out landing."

Tower was advised of this by Mr. TICER and asked if they were hearing him. The tower replied, "It is too late the aircraft has crashed and is burning south of the field."

Omar T. CRABB Approach Controller

Witness: Marion F. TICER

Attachment to strip for Navy 8043:

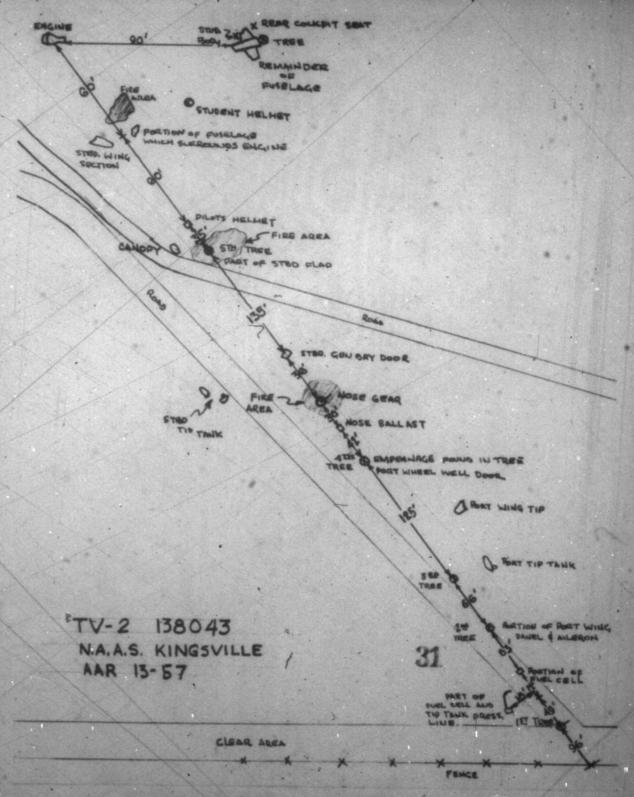
From position in tower cab, the only transmission heard was on 252.4 as follows:

"MAYDAY 8043 flame-out. No choice but night crash landing, am at 800 feet."

John T. EHMAN

Gene A. ROBINSON

Enclosure (4)



ENCL. (7) AAR-13-57 TV-2 BU.#138043 WRECKAGE DIAGRAM.

Engine -O Canopy Empennage 10 Fourth Tree Port tip tank Second Tree Initial Inpact on 380 Heading

ENCL (8) AAR 13-57 TV-2 BU#138043. AERIAL VIEW IMPACT AREA LOOKING NORTHWEST.

Unitial Impact 330 heading for First Tree Third Tree, Second Tree Burned Area Badly Burned Starboard Wing Fuselage. Starboard Wing

ENCL (9) AAR 13-57 TV-2 BU#138043. AERIAL VIEW OF IMPACT AREA LOOKING SOUTHEAST.



ENCL (10) AAR 13-57 TV-2 BU#138043. FIRST TREE - CONTACTED BY PORT WING.



ENCL (10) AAR 13-57 TV-2 BU#138043. FIRST TREE - CONTACTED BY PORT WING.



ENCL (11) AAR 13-57 TV-2 BU#138043. VIEW PROM THIRD TREE LOOKING SOUTHEAST SHOWS SECOND TREE, FIRST TREE, AND INITIAL CONTACT POINT.



ENGL (12) AAR 13-57 TV-2 BU#138043. THIRD TREE - SHOWING SCAR WHERE A/C SHEAVED LARGE LINE PORT TIP AT BASE OF TREE.



ENCL (13) AAR 13-57 TV-2 BU#138043. VIEW FROM THIRD TREE LOOKING NORTHWEST SHOWS FLIGHT PATH AFTER CONTACT OF THIRD TREE.



ENCL (14) AAR 13-57 TV-2 BU#138043. EMPENNAGE AND PORT GUNBAY DOOR HEADING 3300.



ENCL (15) AAR 13-57 TV-2 BU#138043. VIEW OF ENGINE SHOWING VERY LITTLE DAMAGE.



ENCL (16) AAR 13-57 TV-2 BU#138043. REMAINDER OF FUSELAGE.



ENCL (17) AAR 13-57 TV-2 BU#138043. VIEW SHOWING BURNED INTERIOR OF FRONT AND REAR COCKPIT.